


Modeling of Collisionless and Kinetic Effects in Thruster Plasmas

Igor Kaganovich
Princeton Plasma Physics
Laboratory
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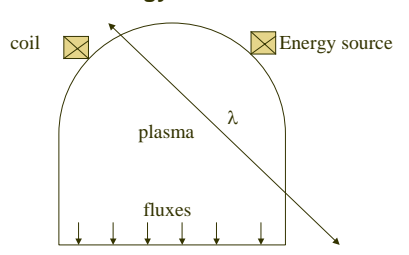


Main Subject of the Talk

Nonlocality and collisionless kinetic effects in low-pressure plasmas.

Nonlocality is important for many plasma applications

- Electron energy mean free path is large, this allows remote plasma handling via nonlocal electron energy distribution function.

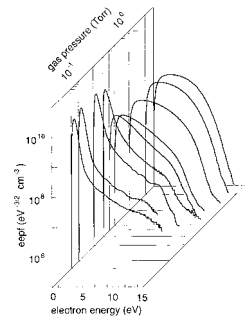


The treatment has to be kinetic! (1/2)

- Most remote from thermodynamic equilibrium:
 - T_e differs from T_i

3eV	$3 \cdot 10^{-2}$ eV	glow discharges
$3 \cdot 10^{-3}$ eV	$3 \cdot 10^{-2}$ eV	Afterglow, plasma jet
100eV	10eV	Hall thrusters
10keV	1eV	ECR ion sources

The treatment has to be kinetic! (2/2)



Electron energy distribution functions are nonMaxwellian:

- Parts of the EDF are very flexible and almost independent.
- Example of capacitive discharge

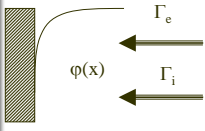
50 Years of History of kinetic studies in low-temperature plasmas

- 1954 - I. Bernstein and T. Holstein
 - Positive column of dc discharge
- 1974 - L. Tseng
 - Positive column of dc discharge
- 1990 - I. Kaganovich
 - RF discharges

For more info: M.A. Lieberman and A.J. Lichtenberg, "Principles of Plasma Discharges and Material Processing", John Wiley & Sons Inc., sec. ed., NY 2005.

Plasma potential relative to wall: kinetic treatment

(1/3)



$$\Gamma_e = n \sqrt{\frac{T_e}{2\pi m}} e^{-\phi/T_e}$$

$$\Gamma_i = n \sqrt{\frac{T_e}{M}}$$

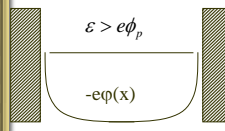
$$\Gamma_e = \Gamma_i \quad e\phi_p = T_e \ln \left(\sqrt{\frac{M}{2\pi m}} \right) \sim 5T_e$$

- Relies on assumption of an Maxwellian electron energy distribution function.
- Is it?

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Plasma potential relative to wall: kinetic treatment

(2/3)

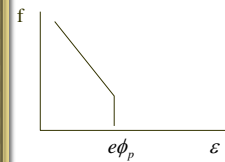


- Most electrons are trapped in potential well.

- Electrons with $\epsilon > e\phi_p$ quickly leave

- \Rightarrow EDF is empty in the loss cone

- \Rightarrow Plasma potential relative wall is smaller

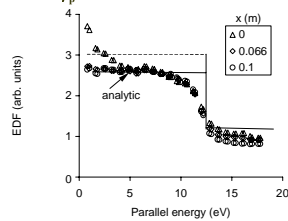


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Plasma potential relative to wall: kinetic treatment

(3/3)

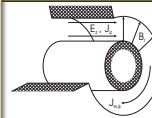
Results of particle-in-cell code for electron energy distribution function (EDF) at given total energy $1.5\phi_p$ as a function of the parallel energy at various coordinates, solid lines correspond to the theoretical estimate at $p=1\text{mTorr}$, $L=10\text{cm}$, $\phi_p=12.4\text{eV}$.



For more info: IEPC-096; Kaganovich, I.D., et al, Phys. Rev. E, **61**, 1875 (2000).

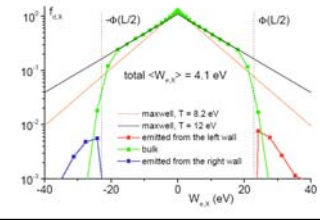
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Kinetic Modeling of Hall Thruster



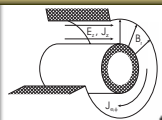
$E_z=52\text{ V/cm}$ $B=100\text{G}$

$T_x=12\text{eV}$ $\phi_p=22\text{eV}$! Not $5T_e$

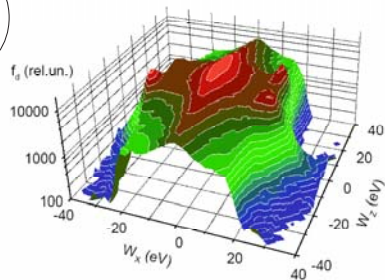


For more info
D. Sydorenko, et al.
IEPC-078 "Session 3-2,
Tuesday, 4pm.
PoP, IEEE TPS (2005)

EEDF is anisotropic + beams due to secondary electron emission



$E_z=200\text{ V/cm}$
 $B=100\text{G}$
 $\langle W_x \rangle = 5.7\text{ eV}$
 $\langle W_z \rangle = 24.5\text{ eV}$



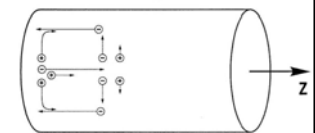
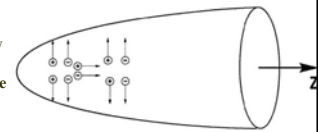
Plasma Expansion Along Magnetic Field Lines

(1/2)

The scheme of non-ambipolar diffusion accompanied by short-circuiting currents. Arrows show the electron and ion motion. The arrow length represents the value of particle fluxes.

(a) Hypothetical ambipolar expansion.

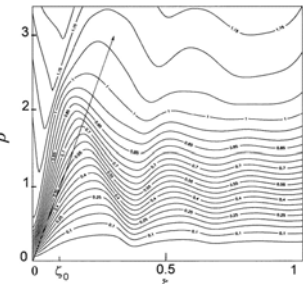
(b) Real non-ambipolar fluxes.



b)

Plasma Expansion Along Magnetic Field Lines (2/2)

Ion trajectory (streamlines of ion flow) in cylindrical geometry. Ion jet propagates with a supersonic velocity along ζ and expands radially due to the electron temperature. The arrow shows the characteristic angle of plasma expansion at the sound velocity.



For more info: IEPC-096; Kaganovich, I.D., et al, Plasma Sources Sci. Technol. 5, 743 (1996).

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Conclusions (1/3)

- Low-pressure plasma has to be treated kinetically
- Fluid approach can lead to quantitatively and qualitatively incorrect results.
- The electron energy distribution function is often non-Maxwellian.
- Electrons tend to stratify into different groups depending on their origin and confinement, i.e., whether they are trapped or not by the plasma potential.
- These different groups of electrons have to be treated separately, can not be lumped together into one Maxwellian EEDF.

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Conclusions (2/3)

- The concept is demonstrated on an example of calculation of particle and heat losses from bounded plasma in presence of strong secondary electron emission as pertains, but not limited to a Hall thruster.
- It was shown that EEDF is nonMaxwellian and strongly anisotropic with $T_x \ll T_z$.
- This strongly influence the plasma potential relative to sheath and particle and heat fluxes to the bounding walls.

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Conclusions (3/3)

- The plasma jet's radial expansion is studied in presence of strong axial magnetic field.
- Electrons can flow fast along the magnetic field lines and simultaneously diffuse radially far from jet injection point and then come back along another magnetic field line. As a result, ions flow freely to the radial position where electrons are available due to such complex loop-like electron trajectories. These short-circuiting flows result in a cylindrical shape of the jet in contrast to conical.
- A practical application of this result can be analysis of differences between testing, laboratory environment and real space environment.

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Related Talks:

- **Y. Raitses,**
 - PPPL
 - IEPC-052, "Electron Temperature Saturation With The Discharge Voltage In Hall Thrusters",
 - Session 5-1, **Wednesday, 5pm.**
- **D. Sydorenko,**
 - University of Saskatchewan
 - IEPC-078 "Kinetic Simulation Of Effects Of Secondary Electron Emission On Electron Temperature In A Hall Thruster",
 - Session 3-2, **Tuesday, 4pm.**

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Is it simple?

- **Plasma is highly**
 - Nonlinear
 - Nonlocal
 - Collisionless
- **Results are often unexpected and surprising**

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Non-monotonic EEDF yields Negative Plasma Conductivity!

EDF afterglow Ar:NF₃, E/N=2 10⁻¹⁷ Vcm², 0, 0.25, 1, 3, 5, 10 ns.
N.A. Dyatko, *et al.*, Plasma Phys. Rep. 1998

$$\mu = -\frac{2e}{3m} \int u^{1/2} \lambda \frac{df(u)}{du} du < 0$$

Total electron flux is directed opposite to the electric field²⁰