



Electron transport in closed $E \times B$ drift devices

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Outline

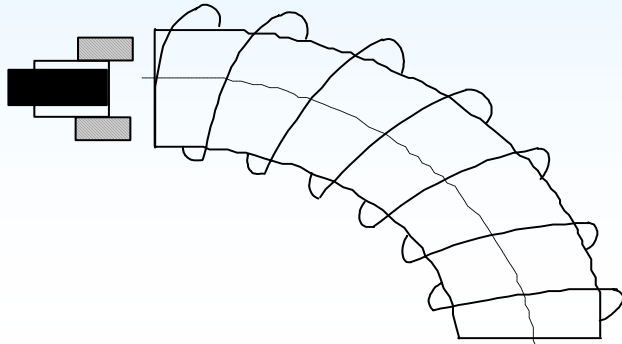
- Plasma devices with $E \times B$ drift
 - PIII with magnetic control
 - Magnetrons
 - Hall thruster
- Electron transport issues overview
- Electron transport model
 - near wall conductivity
- Summary



Plasma immersion ion implantation

Vacuum arc plasma source

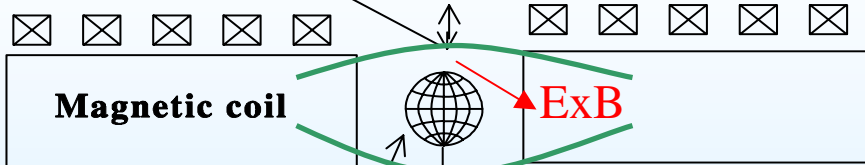
Plasma duct



cathode:	Titanium
arc current:	500 A
arc duration:	250 ms
bias voltage:	2-8 kV
magnetic field:	0-0.12 T

Plasma jet

Movable probe

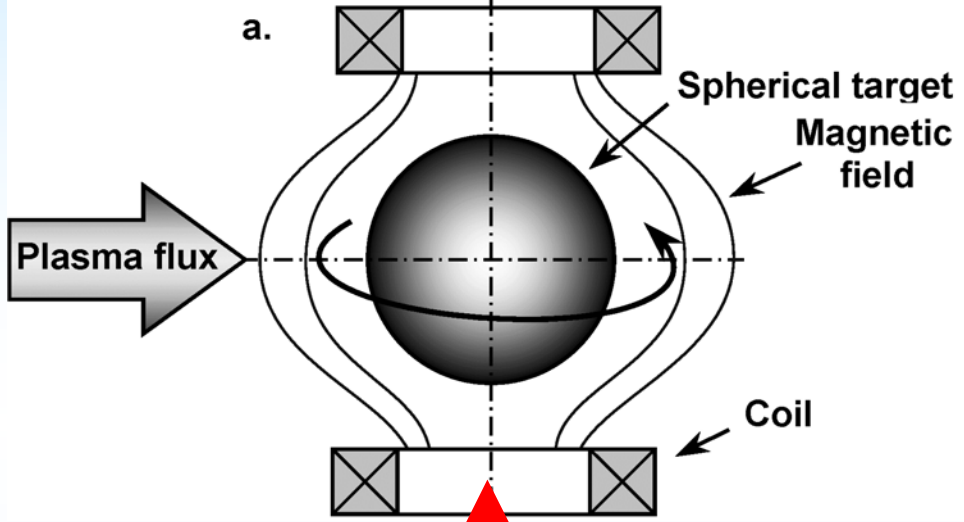


Biased target

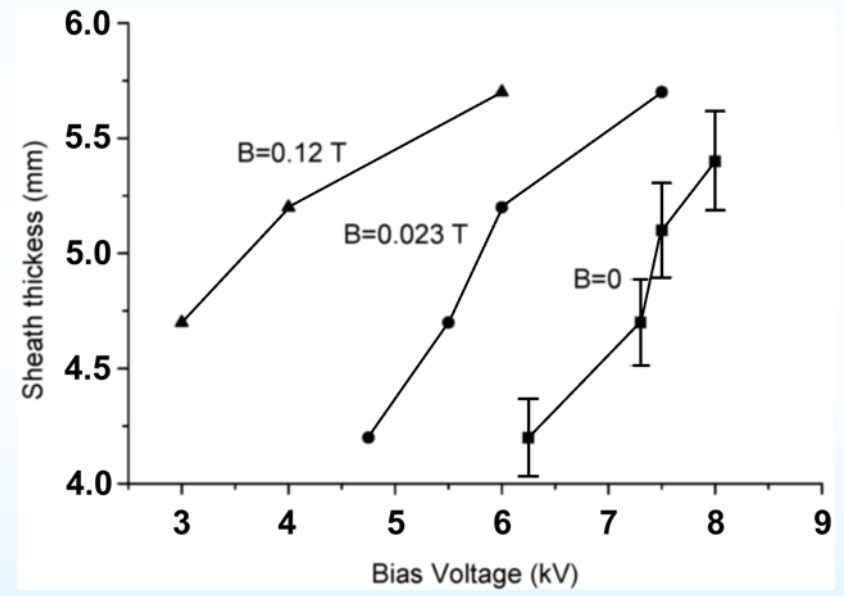
Bias supply



Electron drift



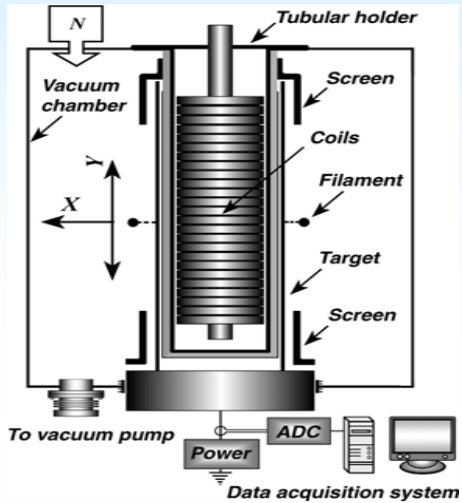
To control sheath
(breakdown vs expansion)



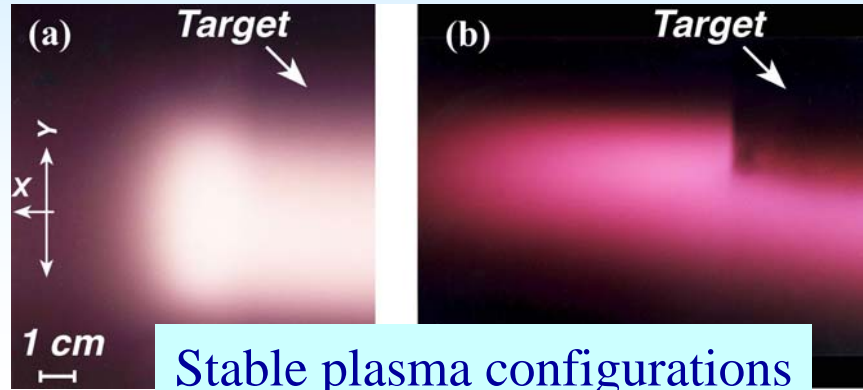
Keidar *et al.* Appl. Phys. Lett, 2002



Magnetrons



Abrupt transition



Stable plasma configurations

Wang & Cohen, 1999

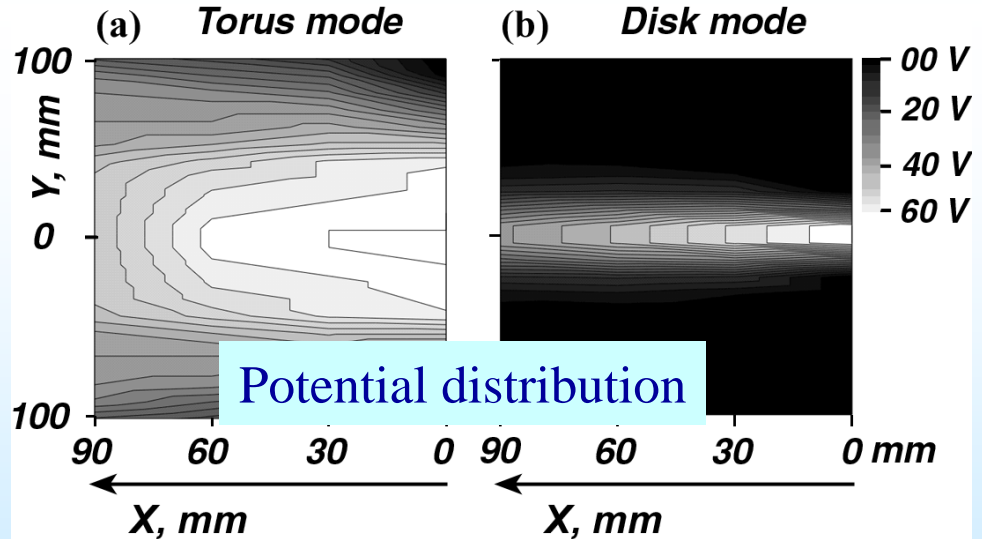
transition
~400 V

$$U_b \gamma_i = \varepsilon_c + \frac{E_{loss}}{\tau_e v_{iz}}$$

Energy balance

$$\frac{1}{v_{iz}} \cong \frac{L}{\mu_B E}$$

Particle balance



Potential distribution

Appl. Phys. Lett, 2004



Magnetrons

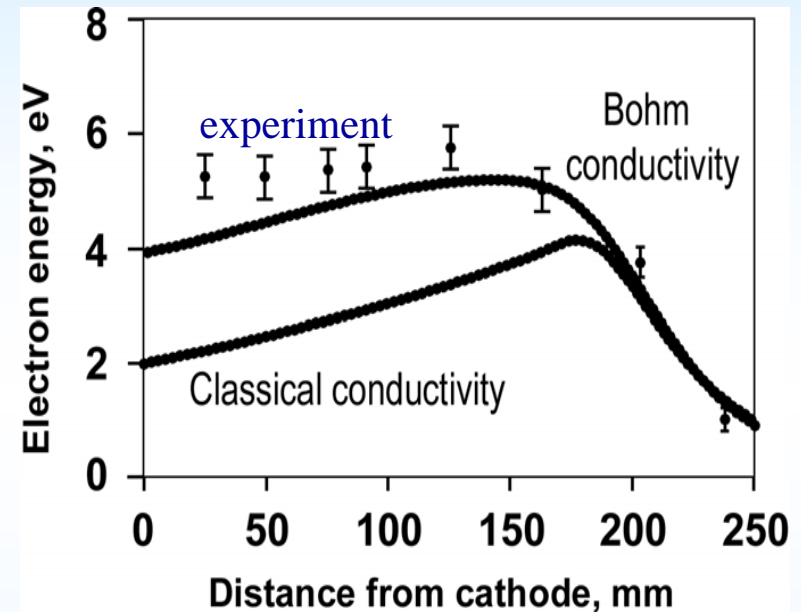
$$\frac{\partial \mathcal{E}_e}{\partial r} = E - \psi_e \mathcal{E}_i$$

$$\mu_{et.B} = \frac{1}{\alpha B}$$

$$\frac{\partial f_i}{\partial r} + \frac{eE}{m_i V_i} \frac{\partial f_i}{\partial V_i} = \frac{n_a}{V_0 V_i} \int_0^\infty \sigma_i V_e f_e dV_e$$

$$j_e(r) = e \cdot n_e \cdot \mu_{et.B} E = \frac{e \cdot n_e \cdot E}{\alpha B}$$

E from experiment

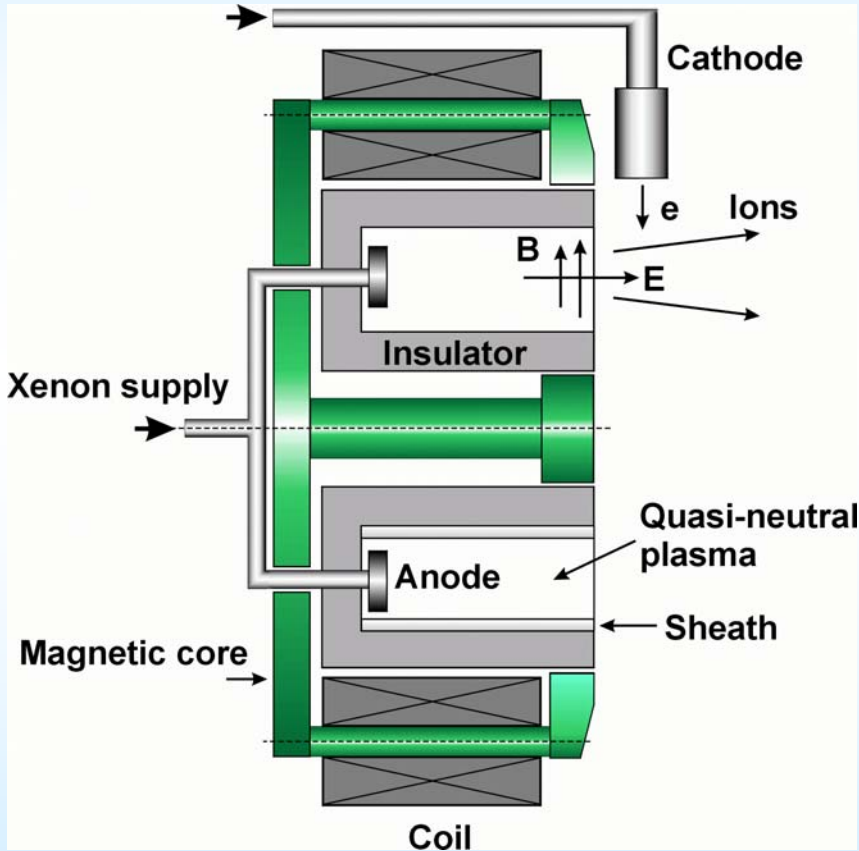


Rosnagel and Kaufman, 1987

$I(\text{classical}) < I_{\text{exp}}$
Plasma oscillations to support
anomalous transport



Hall thruster



$B \sim 0.01 \text{ T}$
 $E \sim 10^4 \text{ V/m}$
 $n_e \sim 10^{17} \text{ m}^{-3}$
 $h \sim 1 \text{ cm}$



Plasma flow modeling approach in Hall thrusters

$$\bullet \quad \partial(nV_z)/\partial z + \partial(nV_r)/\partial r + nV_r/r = \beta n_i n_a$$

$$\bullet \quad V_z \partial V_z / \partial z = - V_r \partial V_z / \partial r + e/m E_z - \beta V_a n_a$$

$$\bullet \quad V_z \partial V_r / \partial z = - V_r \partial V_r / \partial r + E_r$$

$$\bullet \quad j_{er}=0; \quad \varphi - T_e \ln n = \text{const}$$

$$\bullet \quad j_{ez} = en\mu / (1 + (\omega v)^2) (E_z + \partial T_e / \partial z + T_e \partial \ln n / \partial z)$$

$$\bullet \quad 3/2 \partial(j_e T_e) / \partial z = Q_j - Q_w - Q_{ion}$$

2D part

1D part



Electron collisions

Neutrals $\nu_{en} = n_a \sigma_{ea} V_{th}^e$

Bohm $\nu_B = \alpha \omega_e$
($\alpha \sim 1/16$)

wall $\nu_{ew} \sim V_{th}^e / h$

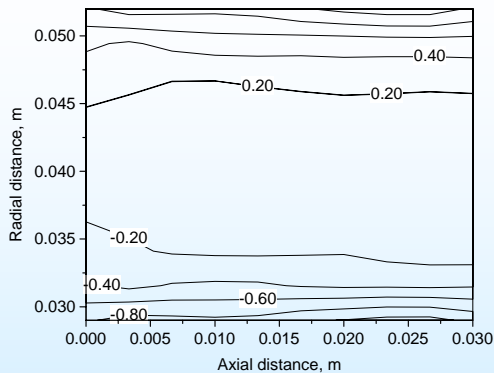
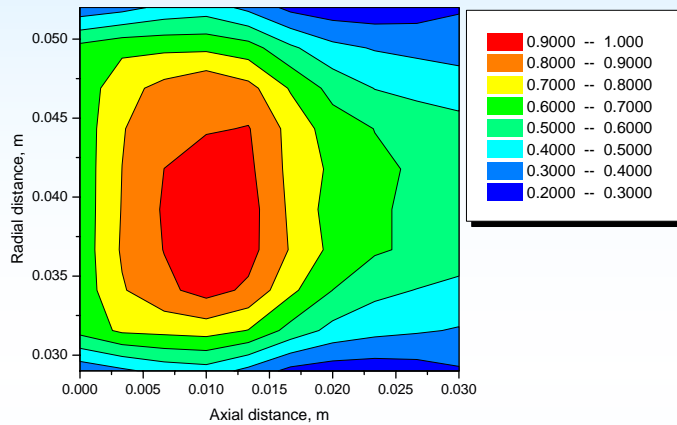
- Bohm-type:
 - Ahedo: 1/80
 - Fife: 1/100
 - Keidar: 1/40
- NWC:
 - Garrigues: $0.2e7 \text{ s}^{-1}$
 - Koo: $(0.2-0.3)e7 \text{ s}^{-1}$

Various models were developed PIC, hybrid, hydrodynamic
All rely on some anomalous coefficient

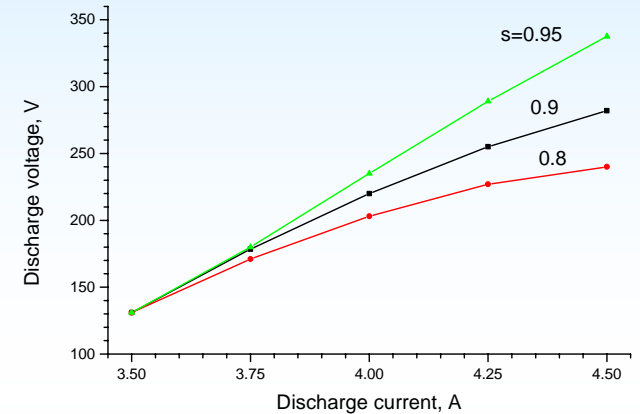
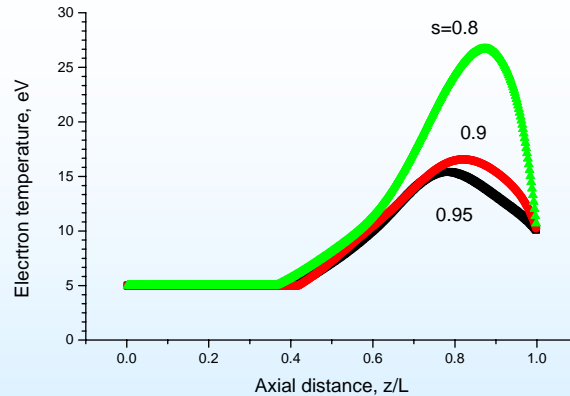


Density & velocity distribution

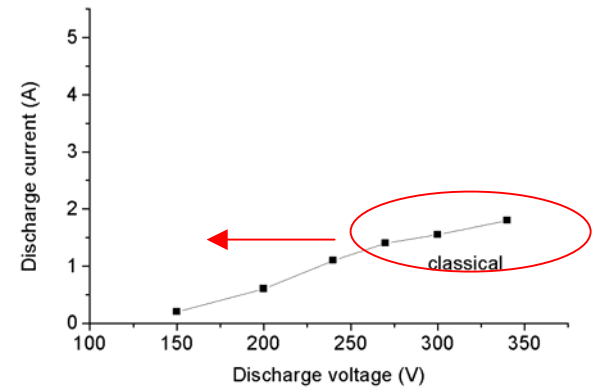
density



Electron energy

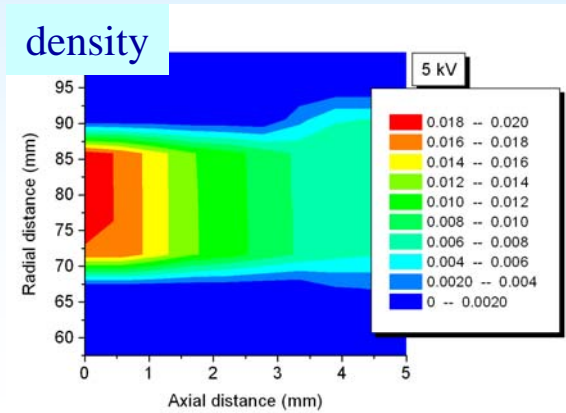
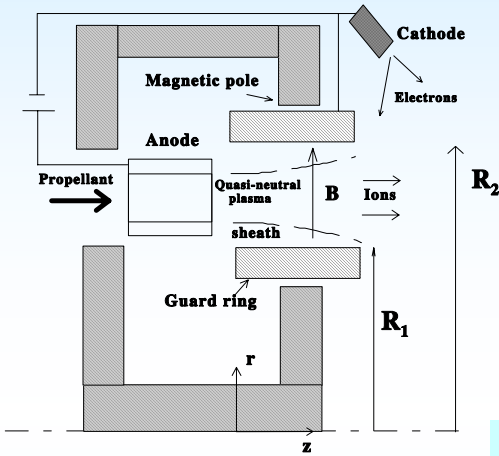


Experiment: $I \sim 4.5$ A, $U_d = 300$ V





Thruster with Anode Layer



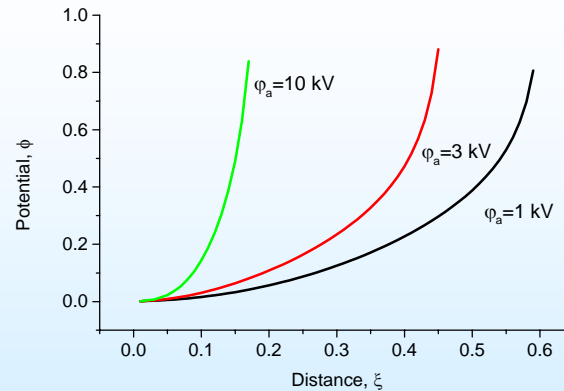
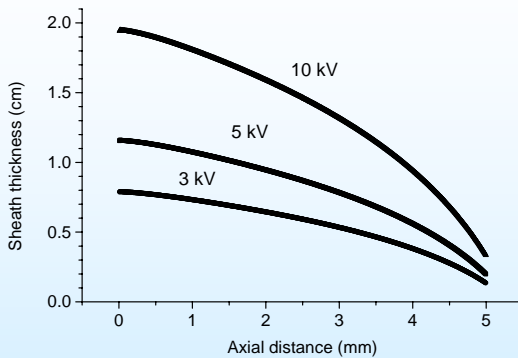
$$e\mu_{\perp}n_e U/d \cong I_e/A(s)$$

$A(s)$ -channel cross section

$$\mu_{\perp} = 1/(\alpha B) \text{ [Bohm]}$$

Plasma detaches from the wall

$$d \sim r_{Ce}A(s)$$



Sheath expansion leads to shorter acceleration region



Modeling assumptions

State-of-the-art

- Electron conductivity mechanism
 - Bohm (plasma turbulence)
 - Near wall conductivity
- Current (potential) distribution
 - “thermalized” potential
 - Uniform current



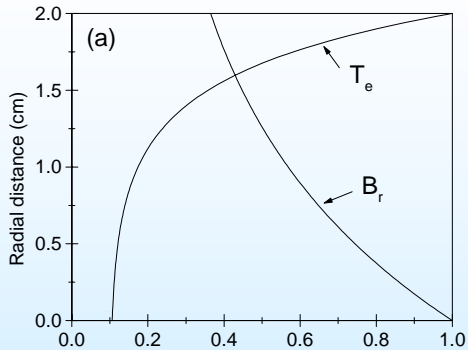
Non-uniform electron transport in HT

current conservation

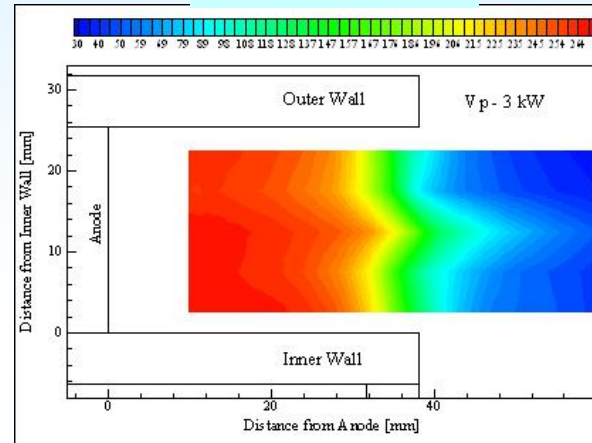
$$j_r = \sigma \left(-\frac{\partial \phi}{\partial r} + \frac{\partial T_e}{\partial r} + T_e \frac{\partial \ln n}{\partial r} \right)$$

$$j_z = \frac{\sigma}{(1 + \beta^2)} \left(-\frac{\partial \phi}{\partial z} + \frac{\partial T_e}{\partial z} + T_e \frac{\partial \ln n}{\partial z} \right)$$

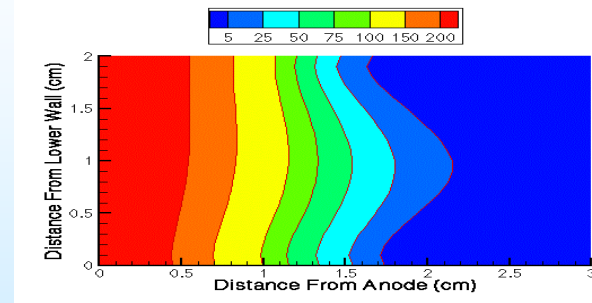
$$\frac{\partial j_r}{\partial r} + \frac{\partial j_z}{\partial z} + \frac{j_r}{r} = 0$$



experiment



simulations

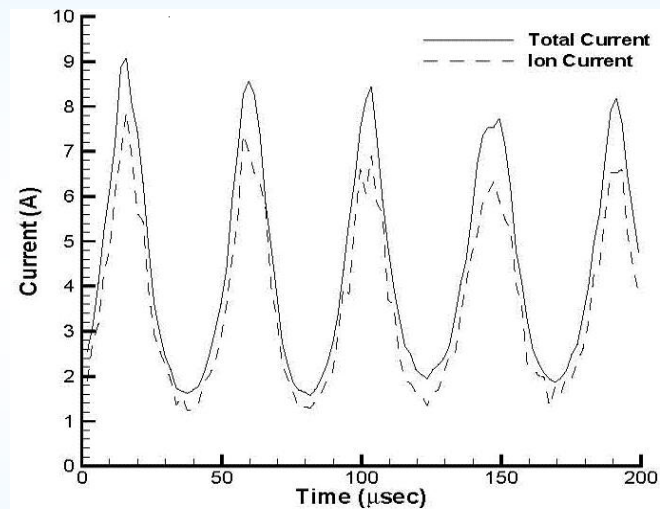




Plasma oscillations

- May be in support of Bohm anomalous mobility
- Experimental evidence in Hall thruster
 - Dependent on mass flow rate, discharge voltage, facility, magnetic field, cathode

2D PIC-MCC





Oscillations

$B \sim 0.01 \text{ T}$
 $E \sim 10^4 \text{ V/m}$
 $n_e \sim 10^{17} \text{ m}^{-3}$
 $h \sim 1 \text{ cm}$

Band	Nature	conditions	Experiment	Simulations
1-20 kHz	Discharge oscillations (contour) Ionization “breathing mode”	PPU	yes	yes
5-25 kHz	Rotating spoke azimuthal	Ionization process	yes	
20-60 Hz	Gradient induced	$dB/dz < 0$	yes	Theory (Morozov)
20-100 kHz	Azimuthal waves	Gradient-driven or ionization	yes	
70-500 kHz	Transient-time Axial electrostatic wave	U_i/L	yes	Hybrid code
0.5-5 MHz	Azimuthal, 5-8 MHz	Rayleigh	yes	Theory (Litvak et al) Gradient magnetic field, density, electron drift velocity



Oscillations

- Drift-dissipative

$$\omega_d = k \left(\frac{T_e}{B} \right) \frac{\langle n \rangle}{n}$$

- Maximum increment

$$\gamma \sim \omega_d \sim \left(\frac{T_e}{B} \right) \frac{\langle n \rangle}{n}$$

- Diffusion

$$D \sim \lambda_{\perp}^2 \gamma$$

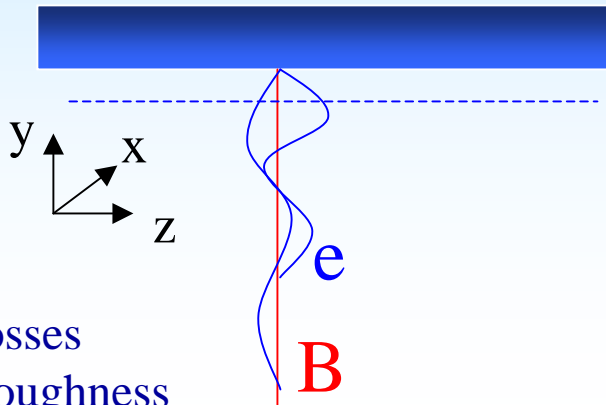
- $D \sim T_e / B$

$$D = \frac{1}{16} \cdot \frac{T_e}{B} \quad \text{Bohm}$$

Alcock & Keen, 1970 Afterglow plasmas
Conditions similar to Hall thrusters



Near wall conductivity



Energy losses
Surface roughness
SEE

Original model

Morozov, 1968-2000

Bugrova, 1985 (experiment)

Recent

Ivanov & Bacal (neutralization)

Barral (sheath effect)

Mean free path: ~ 1 m

Distance between wall: ~ 1 cm



Near wall current

Collisionless kinetic equation

$$\frac{\partial f}{\partial t} + V \frac{\partial f}{\partial r} - \frac{e}{m} (E + V \times B) \frac{\partial f}{\partial v} = 0$$

Emitted electrons

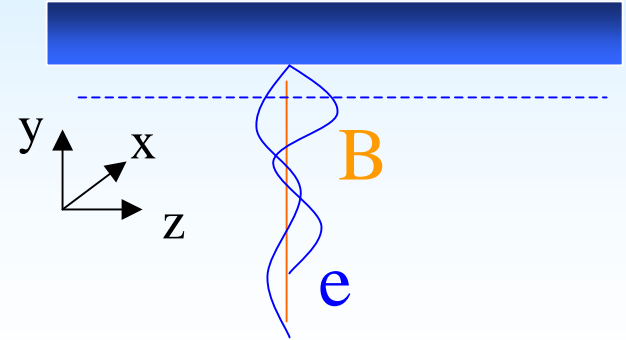
$$f(v) = n_0 \left(\frac{m}{2\pi k T_w} \right)^{3/2} \exp\left(-\frac{mV^2}{2kT_w}\right)$$

Electron dynamics

$$V_{ez} = (V_{ex}^o - V_E) \sin(\omega t) + V_{ez}^o \cos(\omega t)$$

$$V_{ex} = V_E + (V_{ex}^o - V_E) \cos(\omega t) - V_{ez}^o \sin(\omega t)$$

$$V_{ey} = \sqrt{V_{ey}^w + \frac{2e\phi_w}{m}}$$



Characteristics

$$\frac{dV_x}{dt} = -\omega V_z$$

$$\frac{dV_y}{dt} = 0$$

$$\frac{dV_z}{dt} = -\frac{eE}{m} + \omega V_x$$

Electron current

$$j_{ez} = \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} f(v) V_z dV_x dV_z dV_y$$

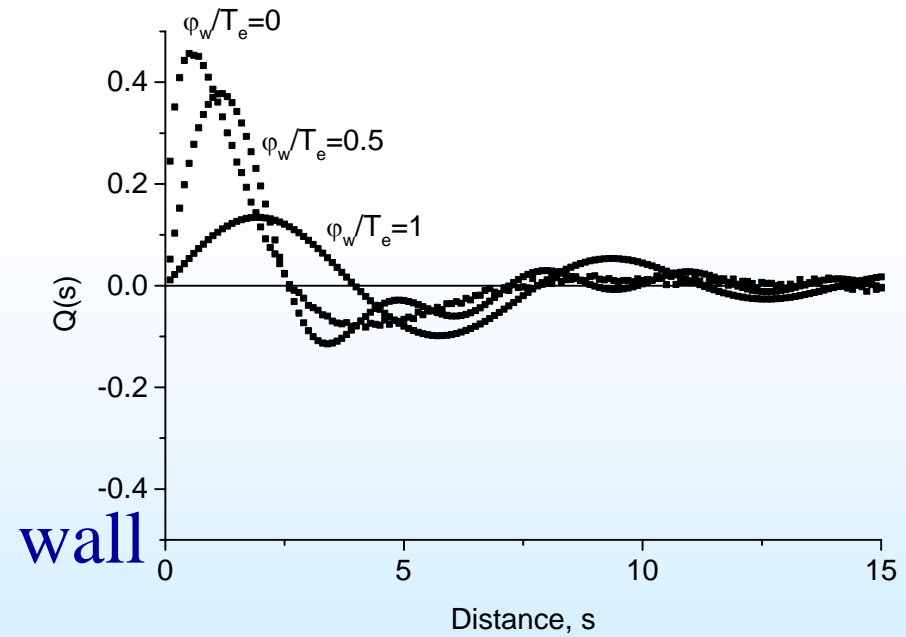


Near wall current

$$j_{ew} = n_0 \frac{E}{B} \left(\frac{m}{2\pi k T_w} \right)^{1/2} \exp\left(\frac{e\phi_z}{kT_w}\right) \exp\left(\frac{e\phi_w}{kT_w}\right) \int_{\sqrt{\frac{e\phi_w}{m}}}^{\infty} \exp\left(-\frac{mV_y^2}{2\pi k T_w}\right) \sin\left(\omega \frac{y}{V_y}\right) dV_y$$

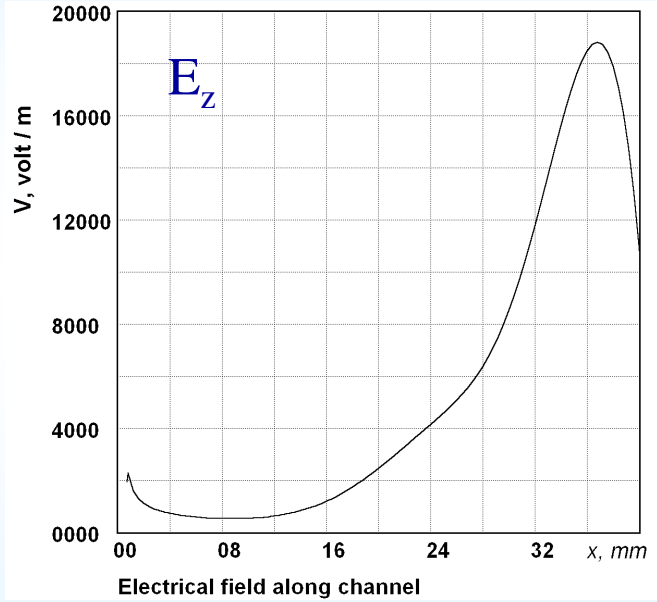
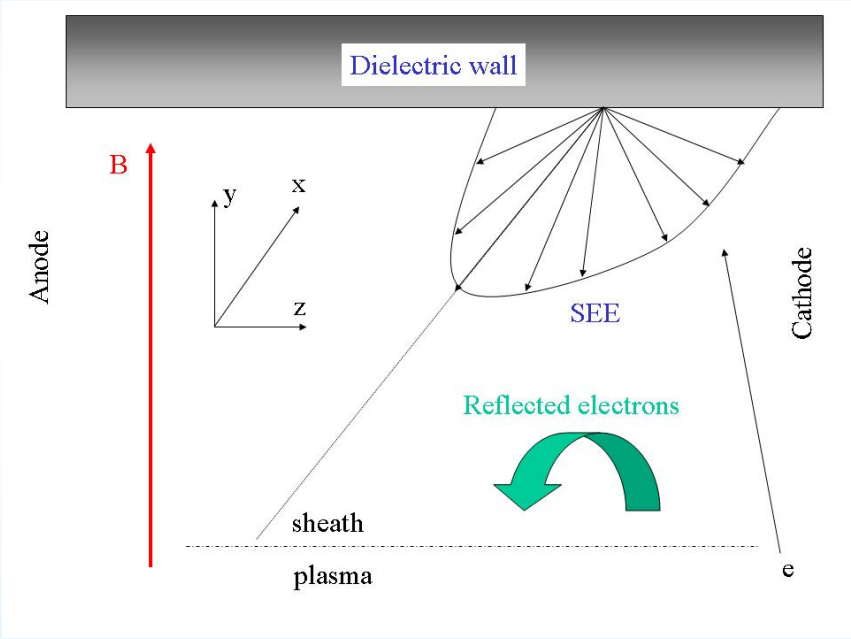
$$Q(s) = \int_{\sqrt{\frac{e\phi_w}{m}}}^{\infty} \exp(-\alpha^2) \sin\left(\frac{s}{\alpha}\right) d\alpha$$

$$\alpha = \frac{V_y}{\sqrt{\frac{2kT_w}{m}}} \quad s = \frac{\omega y}{\alpha \sqrt{\frac{2kT_w}{m}}}$$





Axial electric field effect on SEE

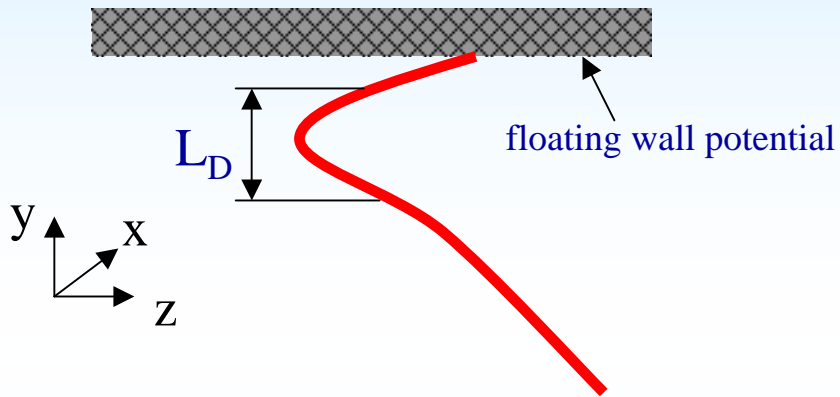


In sheath: $E_z \sim 10^4$ V/m; $E_r \sim 10^5$ V/m



Axial E-field. Current enhancement

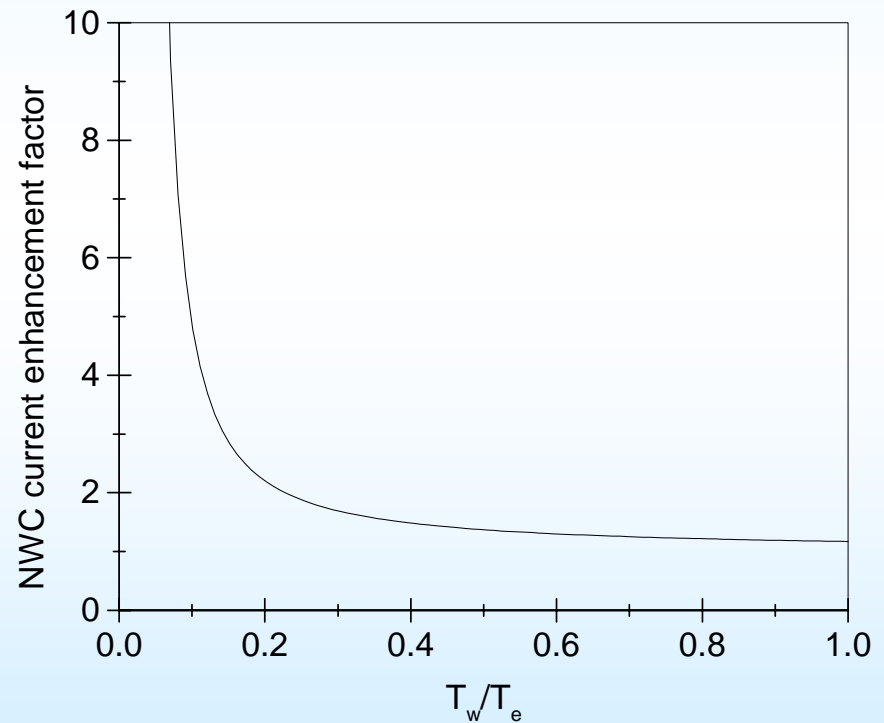
space charge saturated sheath



$$\Delta\phi_z = E_z^2 \int_{y_{\min}}^{y_{\max}} \frac{dy}{E_y(y)} \approx E_z^2 \frac{L_D}{E_y}$$

$$\Delta\phi_z \approx E_z L_D$$

NWC depends on wall to bulk T_e ratio





Summary

- PIII, magnetrons, Hall thrusters have similar physics, ExB drift. This leads to efficient ionization and high electric field in the quasi-neutral plasma region
- Electron transport is largely determined by properties of the drift region (walls, plasma oscillation)
- Anomalous transport mechanism is needed to explain electron current in most cases (Bohm, NWC)
- Conditions for both mechanisms are different and required further evaluation for each specific device